



SPEED CONTROL OF A MICROPROCESSOR BASED DC DRIVE

Part I: Design of the Control Loops

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ABSTRACT

The system under consideration is a DC motor drive system consisting of a separately excited DC machine supplied from a DC-chopper and controlled through a microprocessor trainer. In this part of the study, the suitable control loops and their digital controllers are designed. The study is continued in a second part (Part II) of this article where the complete structures of the hardware and the software are discussed.

It is desirable to optimize the efficiency of the overall system and at the same time to obtain a satisfactory transient response of the speed. In the method used, the ratio of the armature current and the field current (I_a/I_f) that gives the maximum efficiency of the DC motor is analytically derived and the error term of the current ratio is controlled to be zero at steady state speed of the motor. However at the transient state, the efficiency of the proposed method is not optimal, but the satisfactory transient response is expected in compensation for the non-optimal transient efficiency. This characteristics is desirable, because the instantaneous value of speed is important. On the other hand, the integral value of efficiency is more important than its instantaneous value.

I. INTRODUCTION

The functional block diagram of studied system is given in Fig.1. The power circuits consist of two DC choppers. The four quadrant DC chopper of the armature circuit consists of two MOSFET switches operating at high chopping frequency, and two darlington power transistors operating in the continuous current conduction mode. This arrangement represents the optimal utilization of the characteristics of both types of devices; fast switching MOSFET and lower conduction losses transistors. Also this arrangement decreases the cost to much. The field circuit consists of a single MOSFET switch constructing one quadrant DC chopper.

In a previous paper [4] the complete design of the two power circuits was designed including selection of the devices, calculation of the power losses and the heat sink requirements. The overall efficiency and its dependence on the load current, the load voltage, the supply voltage and the operating frequency was also established. The active protective means for all devices against the harmful conditions were implemented. The study is continued in

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the present paper where the control circuits are designed to get optimal efficiency. In the second paper (Part II) the complete design of the software and the hardware is given and the experimental results of the whole system are illustrated.

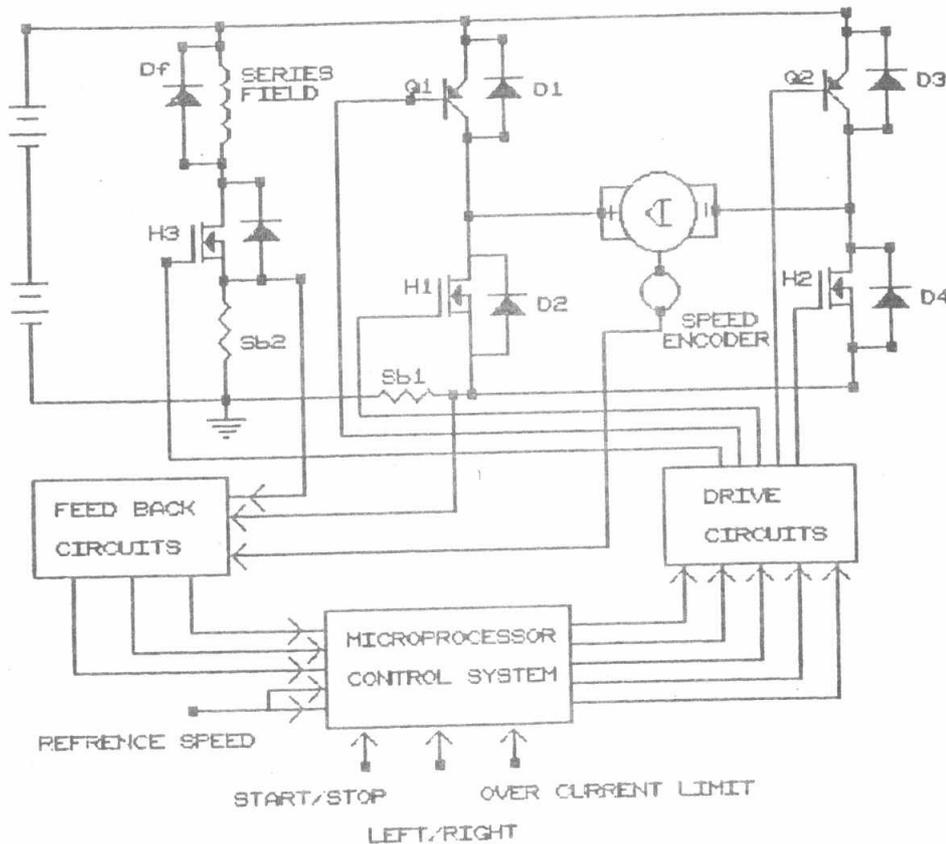


Fig.1 The Function Block Diagram of the Drive System

II. THE ANALYSIS OF THE CONTROL SYSTEM

As stated above, in the present work, it is desirable to optimize the efficiency of a DC drive and at the same time obtaining a satisfactory transient response of the speed. The motor used is a separately excited DC motor. Until now, this motor has been used in the industrial field when high accurate speed control is necessary over a wide speed range. The possibility of the speed control of the system that gives the optimal efficiency by means of controlling both armature and field currents simultaneously has been proposed in several works. However details of the realization of such systems are not considered. In the following section the losses of the machine are studied to clarify the method of control.

A. The Consideration of Losses:

Generally, the efficiency of the DC motor system under a given output power P_o is represented by:

$$n = \left[\frac{P_o}{P_o + P_L + P_m} \right] \quad (1)$$

where: P_m = mechanical losses
 P_L = iron losses + copper losses + stray losses + converter losses

In order to maximize the efficiency (n), the sum of losses must be minimized at the output P_o . The mechanical losses P_m are not controllable under condition of a given desired speed. The controllable losses P_L take the following form:

$$P_L = P_a + P_f \quad (2)$$

where:

P_a = power losses associated with armature current including converter losses. These can be approximated with a good accuracy by:

$$P_a = K_a \times I_a^2 \quad (3)$$

To get the best possible value of K_a , P_a is determined at several values of I_a and least squares method of approximation is used. P_f is the power losses associated with the field current including the converter losses.

$$P_f = I_f^2 R_f + \text{field converter losses} + P_e + P_h \quad (4)$$

The eddy current P_e and the hysteresis P_h losses can be noted in the following:

$$P_e = K_e n^2 B_m^2 \quad \text{and} \quad P_h = K_h n B_m^{1.6} \quad (5)$$

Since B_m depends on I_f , thus P_f is a function of the current I_f and the speed n . If saturation effect is neglected P_f can be approximated with a good degree of accuracy into the following form:

$$P_f = K_f(n) I_f^2 \quad (6)$$

$K_f(n)$ depends on the motor, and usually the linearized value $K_f(N_o)$ at an operating speed N_o is used. Its value is changed at several operating points to cover the wide range of the speed. The suitable values are determined by the same method used to obtain K_a but with the speed as a parameter.

$K_f(N1)$	for speeds up to	1000	rpm.
$K_f(N2)$	for speeds between	1000 and 1500	rpm.
$K_f(N3)$	for speeds between	1500 and 2000	rpm.
$K_f(N4)$	for speeds above	2000	rpm.

Assuming an unsaturated magnetic circuit, the generated torque T_m is given by:

$$T_m = K_T I_a I_f \quad (7)$$

The optimal ratio of I_a/I_f , giving the minimum losses under the constant

generated torque at $n = N_0$ is obtained by means of the Lagrange multiplier method. Using Eq.2, the minimum losses are obtained if $dP_L = 0$ where:

$$dP_L = \left[\frac{\partial P_L}{\partial I_a} \right] dI_a + \left[\frac{\partial P_L}{\partial I_f} \right] dI_f \quad (8)$$

$$\text{Then: } \left[\frac{dI_a}{dI_f} \right] = - \left[\frac{K_f(N_0)I_f}{K_a I_a} \right]$$

Since the torque is constant, then from Eq.7, one gets:

$$dT_m = K_T I_a dI_f + K_T I_f dI_a = 0 \quad (9)$$

$$\text{or } \left[\frac{dI_a}{dI_f} \right] = - \left[\frac{I_f}{I_a} \right]$$

From Eqs.8 and 9, one obtains:

$$\left[\frac{I_a}{I_f} \right]^2 = \left[\frac{K_f(N_0)}{K_a} \right]$$

$$\text{Then: } \left[\frac{I_a}{I_f} \right] = \left[\frac{K_f(N_0)}{K_a} \right]^{1/2} = k(N_0) \quad (10)$$

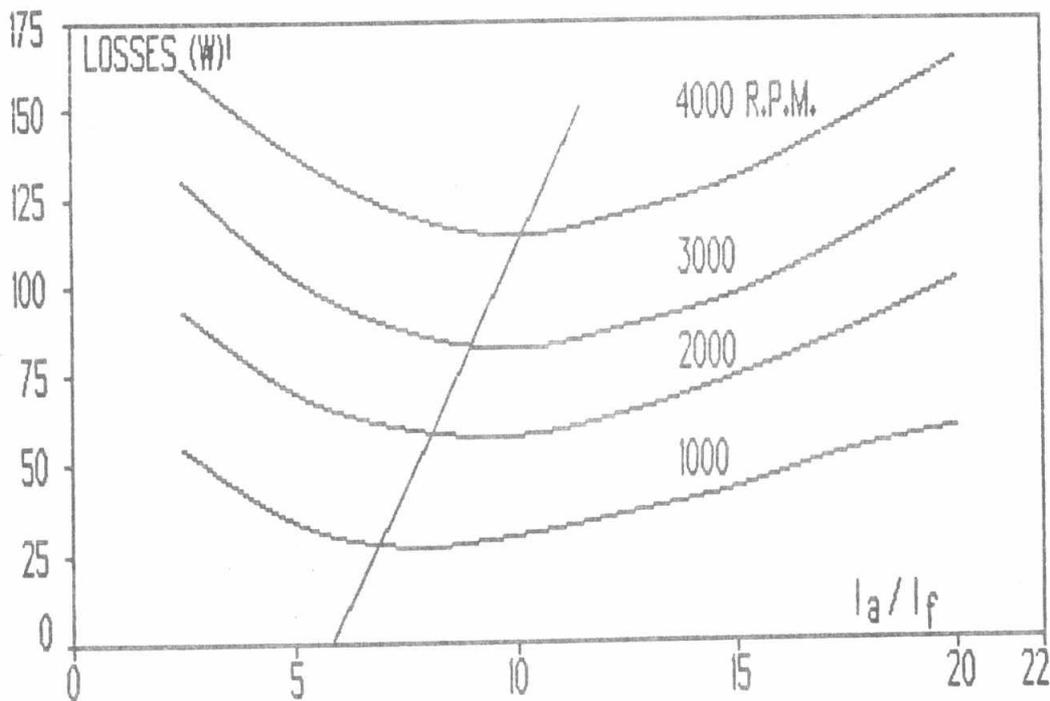


Fig.2 Evaluation of the Speed Constant $k(N)$

The optimal current ratio $k(N)$ defined by Eq.10 is not easy to be obtained by measuring K_f and K_a due to the practical difficulties in separation of armature and field losses with the available test equipments. Instead the motor is tested at a light load where the total input power is only sufficient to cover the total losses. These losses are measured at variable I_a/I_f ratio and constant speed. The results are plotted in Fig.2 at four constant speeds: 1000, 2000, 3000 and 4000 rpm where each graph has a

minimum losses point. From these graphs the following values of $k(N)$ are obtained:

$k(N1)$	= 0.78	= 00.C8 ₁₆	at speed 1000 rpm.
$k(N2)$	= 0.91	= 00.E9 ₁₆	2000 rpm.
$k(N3)$	= 1.05	= 01.0D ₁₆	3000 rpm.
$k(N4)$	= 1.12	= 01.1F ₁₆	4000 rpm.

These values of $k(N)$ are represented internally in the MPU look-up table.

B. Control System Structure:

The control system depicted in Fig.3 consists of three loops; one outer speed loop, and an inner armature current loop and a field current loop.

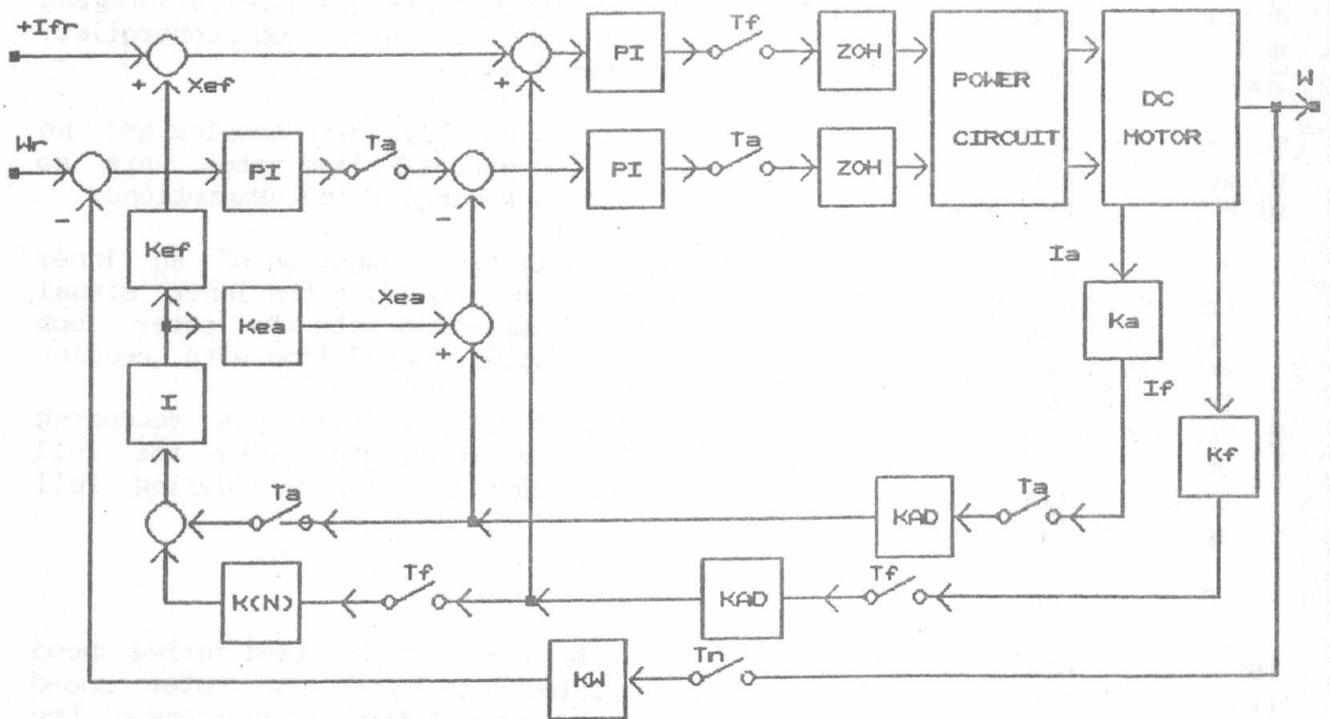


Fig.3 Structure of the Control System

As shown, the control system contains an error term of the current ratio in the form:

$$e_{eff} = I_a - k(N_0)I_f \tag{11}$$

This error is subjected to an integral correction and is used to adjust I_a and I_f to optimize the efficiency. Therefore at steady state this error term becomes zero and optimal efficiency is obtained.

The band-widths of the three control loops are widely different. The highest

frequency signal will determine the sampling frequency. Then to satisfy the sampling theory [5], the sampling period must be smaller than $T_a/2$. With microprocessors of limited computing power as the used M6800 this sampling period is very small to implement the controllers programs and the associated input/output operations.

In the present system, according to its parameters, there are three sampling rates: the armature current is sampled with a sampling period $T_a=1.178$ ms, the field current with a sampling period $T_f=8.245$ ms and the speed with a sampling period $T_n=24.736$ ms. The speed sampling period is dictated by the speed transducer, it is considerably smaller than the mechanical time constant ($\tau_m = 1s$). This is important from point of view of stability of the control system. The sampling periods T_f and T_a are selected also to be sufficiently smaller than τ_f and τ_a respectively and at the same time to be integral divisors of T_n ($T_f=T_n/3$ and $T_a=T_n/21$). This is necessary so that synchronization of different samplers can be possible.

The armature sampling period T_a is the smallest unit of time in the system, during this period the MPU executes the armature current controller program and parts of the field current controller and of the speed controller. Therefore T_a cannot be smaller than the stated value.

The analysis of a multirate digital systems is usually quite complex and no unique approach can be used. Fortunately, the used sampling rates are so widely apart that we can introduce the following simplifying assumptions:

1. In analyzing the outer slower loop, the transfer function of an inner faster loop can be taken as a constant. This is because the inner signal reaches steady state in short period in comparison with the outer loop sampling period. This is true when the analyzing speed loop with respect to field and armature circuits.
2. Concerning the inner armature current loop, the coefficients depending upon field current (K_e and K_f) are taken as constants during the full field sampling period. Also the speed is taken as a constant during full speed sampling period.

C. Transient Response

The control system is designed such that field current is fixed during speed transients, thus control system is simplified into two loops: outer speed loop and inner armature current loop. The s-domain transfer functions of the speed (W) and armature current (I_a) are obtained. At a constant field current the following equations are given:

$$v_a(t) = E + i_a(t)R_a + L_a \left[\frac{di_a(t)}{dt} \right] \quad (12)$$

$$K_T i_a = J \left[\frac{dW}{dt} \right] + F \times W \quad (13)$$

Using these two equations, the block diagram in Fig.4.a is obtained. This can be simplified into the block diagram in Fig.4.b where:

$$G1(s) = \left[\frac{sJ + F}{s^2 L_a J + s(R_a J + L_a F) + (R_a F + K_T K_E)} \right]$$

$$= \left[\frac{1}{L_a} \right] \times \left[\frac{s + (F/J)}{s^2 + s(1/\tau_m + 1/\tau_a) + (1/\tau_m \tau_a + K_E K_T / L_a J)} \right] \quad (14)$$

$$G_2(s) = \left[\frac{K_T}{sJ + F} \right] = \left[\frac{K_T/J}{s + 1/\tau_m} \right] \quad (15)$$

where: $R_a = 0.3 \text{ Ohm}$ $L_a = 0.81 \text{ mH}$
 $F = 8 \times 10^{-4} \text{ Nm/rad.s}^{-1}$ $J = 8 \times 10^{-4} \text{ kg.m}^2$
 $K_T = 0.02 \text{ Nm/A}$ $K_E = 0.02 \text{ V/rad.s}^{-1}$

(At shunt field current = 1.8 A)

$\tau_a = L_a/R_a = 2.7 \text{ ms}$ armature circuit time constant.

$\tau_m = J/F = 1.0 \text{ s}$ mechanical time constant.

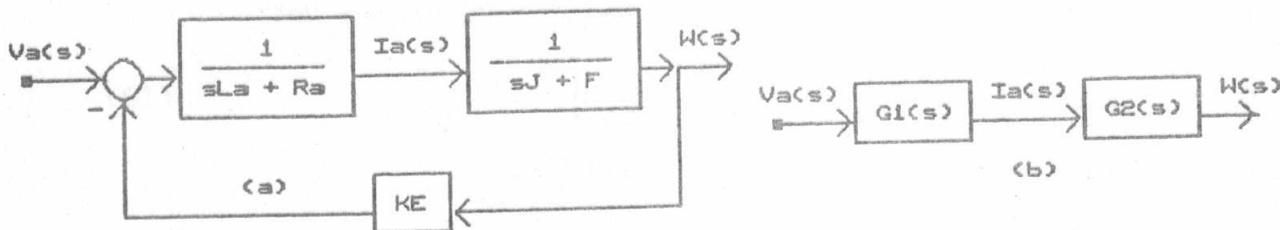


Fig.4 The Armature Controlled DC Drive

III. THE DIGITAL CONTROLLERS

The analysis of the control system will be carried out using the z-transform which is the suitable approach in such digital control system.

A. The armature Current Controller

The microprocessor control system executes a PI correction on the current error e_a such that the armature current I_a follows the reference armature current I_{ra} with zero steady state error. The control algorithm also limits the value of I_a to protect power circuit from over current at abrupt speed variations. In the transient periods this controller has only the single input I_a , but at steady state when efficiency optimization loop is in operation another input (X_a) is presented as shown in Fig.5. The MPU implements the PI controller $G_c(z)$ numerically as:

$$G_c(z) = K_p + \left[\frac{K_i T_i z}{z - 1} \right] = K_d \left[\frac{z - z_d}{z - 1} \right] \quad (16)$$

where: $K_d = K_p + K_i T_i$ and $z_d = K_p / K_d$ (17)

From the block-diagram of the armature current controller shown in Fig.5. the closed loop transfer-function is obtain in the following form:

$$G_a(z) = Z \left[\frac{1 - e^{-sT_a}}{s} G_1(s) \right] = (1 - z^{-1}) Z \left[G_1(s)/s \right] \quad (18)$$

$$G_a(z) = K_{aa} \left[\frac{z - z_1}{(z - p_1)(z - p_2)} \right]$$

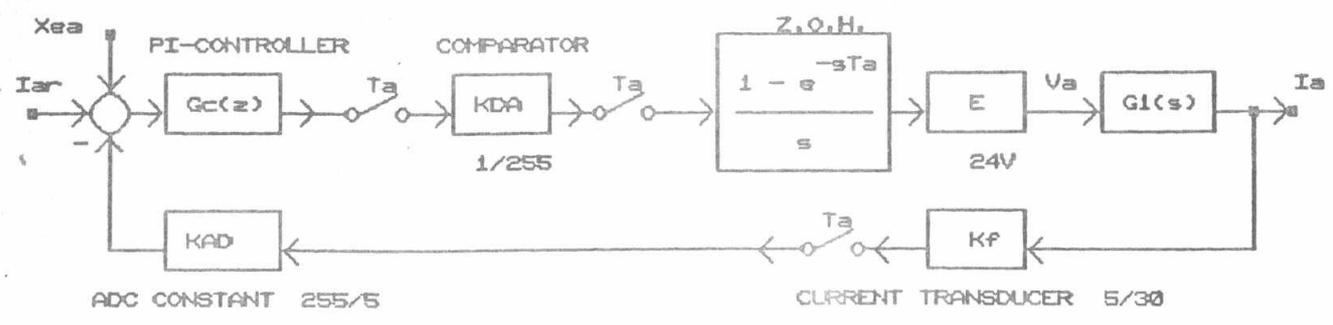


Fig.5 The Block Diagram of the Armature Current Controller

Substituting numerical values and taking the z-transform one get:

$$G_a(z) = 1.1784 \left[\frac{z - 0.9989}{(z - 368.7)(z - 2.68)} \right] \quad (19)$$

The closed loop transfer function, $G_a(z)$ will be:

$$C_a(z) = \left[\frac{K_{AD} E G_C(z) G(z)}{1 + K_{DA} K_f K_{AD} E G_C(z) G(z)} \right] \quad (20)$$

$$= K_a \cdot \left[\frac{z^2 - (z_d + z_1)z - z^1 \cdot z_d}{A_3 z^3 + A_2 z^2 + A_1 z + A_0} \right] \quad (21)$$

where:

$$A_3 = 1 \qquad A_2 = -K_1 + K_2 \cdot K_{da}$$

$$A_1 = K_3 - K_2 \cdot (z_1 - z_{da}) \cdot K_{da} \qquad A_0 = -K_4 + z_{da} \cdot K_{da}$$

Applying Jury's conditions of stability to characteristic Eq.20, then:

1. $F(1) > 0$ then $A_3 + A_2 + A_1 + A_0 > 0$
 this gives: $K_{da}(1 - z_{da}) > 0$ (22)

2. $F(-1) < 0$ then $-A_3 + A_2 - A_1 + A_0 < 0$
 This gives $K_{da} + K_{da} \cdot z_{da} < \dots$ (23)

3. $|A_0| < A_3$
 this gives: $\dots < K_{da} \cdot z_{da} < \dots$ (24)

4. $|A_0^2 - A_3^2| > (A_0 A_2 - A_3 A_1)$ (25)

Equations 22 through 25 give the boundary of K_{pa} and K_{ia} that render stable closed loop system.

B. Speed Controller

The mechanical time constant ($\tau_m = 1s$) of the system is very high compared with the armature time constant ($\tau_a = 2.7 \text{ mS}$) and similarly the speed sampling period T_n is very high in comparison to the armature sampling period T_a . Thus it is a reasonable assumption to take the armature current controller as a constant when analyzing the speed controller. In fact this constant is the scaling factor of the armature current, that is an armature current of 30 A corresponding to the hexadecimal number FF (255₁₀). This armature transfer function in the present case is taken as 30/255. Thus the speed control system is reduced into Fig.6, where:

$$D_n(z) = K_{pn} + K_{in}T_{ni} \left[\frac{z}{z-1} \right] = K_{dn} \left[\frac{z - z_{dn}}{z-1} \right] \quad (26)$$

where: $K_{dn} = K_{pn} + K_{in}T_{in}$ and $z_{dn} = K_{pn}/K_{dn}$

The z-transform of the forward path is obtained in the following:

$$G_n(z) = \left[\frac{30}{255} \right] \cdot G_n(z) (1 - z^{-1}) Z \left[\frac{G_2(s)}{s} \right] \quad (27)$$

or

$$G_n(z) = 0.719 K_{dn} \left[\frac{z - z_{dn}}{(z-1)(z-0.9756)} \right] \quad (28)$$

The closed loop transfer function $C_n(z)$ is obtained as:

$$C_n(z) = \left[\frac{G_n(z)}{1 - G_n(z)K_n} \right] \quad (29)$$

$$C_n(z) = \left[\frac{0.719 K_{dn} (z - z_{dn})}{z^2 - (1.9756 - 0.0719 K_{dn})z + 0.9756 - 0.0719 K_{dn}z_{dn}} \right] \quad (30)$$

Applying Jury's stability conditions to the characteristic equation we obtain the following boundary of K_{pn} and $K_{in}T_{in}$ to obtain a stable closed loop system:

$$27.06 > K_{pn} > 0.3394 \quad \text{and} \quad 2K_{pn} + K_{in}T_{in} > 54.95 \quad (31)$$

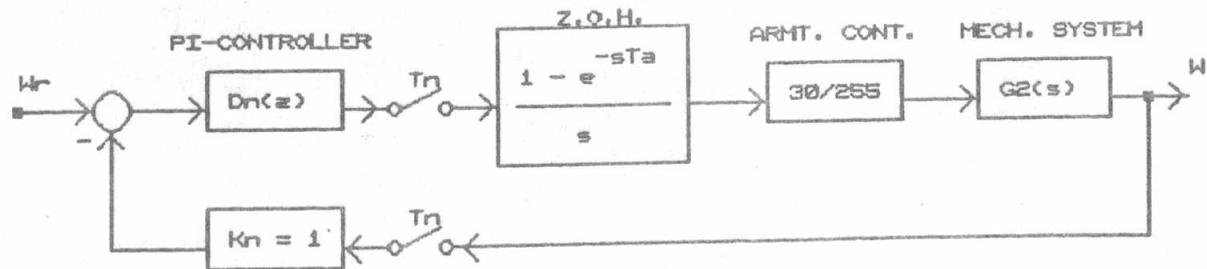


Fig.6 The Speed Controller

Measurement of the Actual Speed (Fig.7)

In the present work the speed is measured by a digital speed transducer as shown in Fig.7. It depends on an optical pulse tachogenerator having a slotted disk coupled with the motor shaft. The tacho-pulses are counted over a sampling period $T_n = 24.736$ ms. The slotted disk has 127 slots on its periphery, and two opto-interrupters are used to the double number of pulses per revolution. The speed sampling time T_n is calculated such that, count expresses the speed in rad/s. Thus if the number accumulated in speed counter at end of sampling period T_n is N the following relation can be written:

$$N = \frac{W \times T_n \times (127 \times 2)}{2 \times 3.1416} \quad (32)$$

If $N = W$, then:

$$T_n = \frac{2 \times 3.1416}{2 \times 127} = 24.736 \times 10^{-3} \quad (33)$$

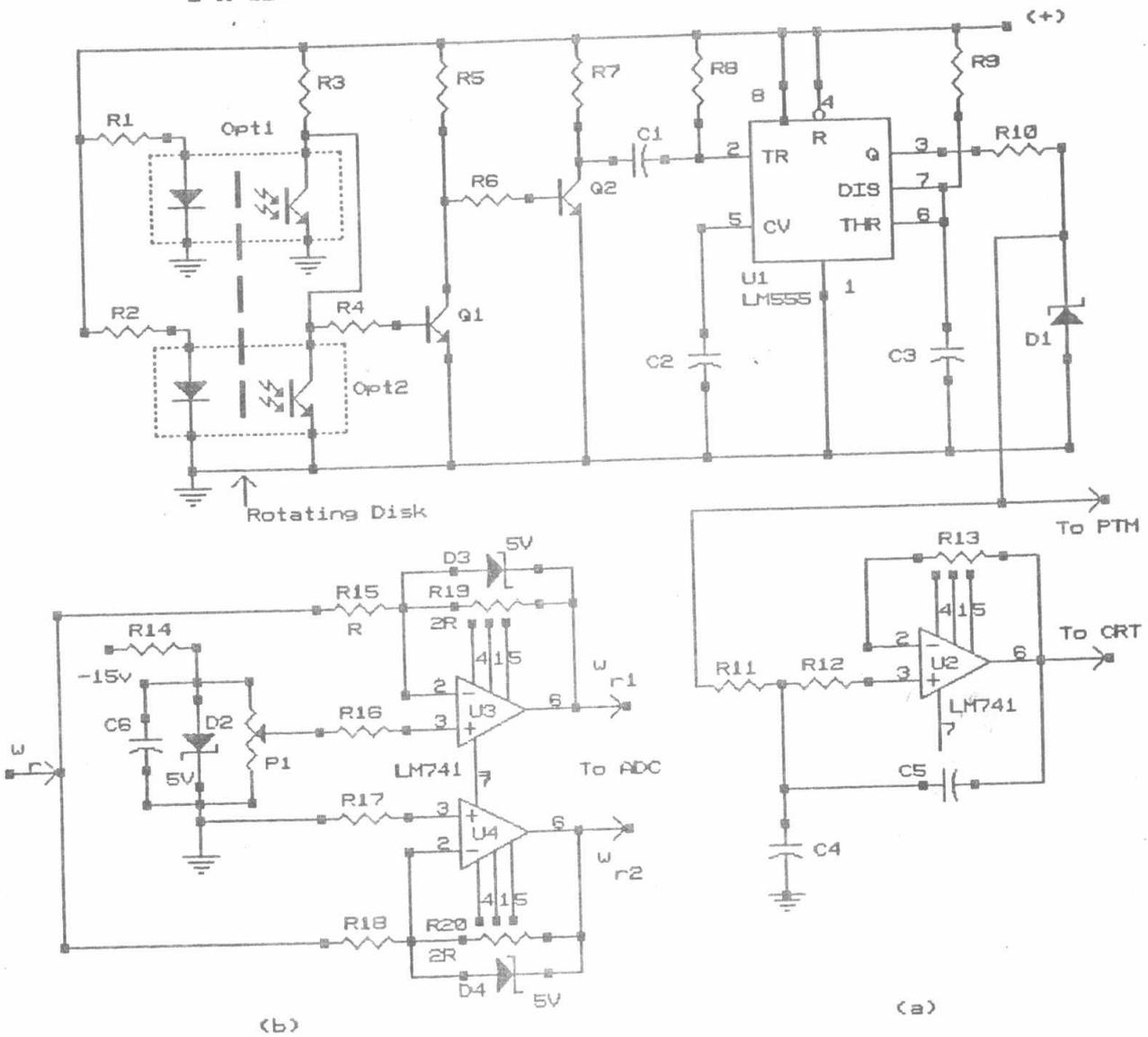


Fig.7 The Speed Measurement

Measuring of the Reference Speed

In present application speed reference is entered via the ADC as an analog signal. The drawback of this method is that the speed measurement resolution can be limited to 1/256 which is the usual resolution of an 8-bit ADC. The circuit shown in Fig.7 together with the software used in Part II of this article improves this resolution to 1/512. The MPU reads the two signals W1 and W2 then it adds the two digital values to obtain the reference speed as a 9-bit number.

C. The Field Controller

The field controller operates independently of the other controllers before abrupting speed transients to set the field current to a value suitable to

such transients. Apart from the speed transients the field controller operates normally to adjust the field current to a value which optimize the efficiency, by adding the input X_{ef} to I_{fr} as shown in the control block

diagram of Fig.1. It is to be noted that I_{fr} is a constant predetermined value and not an output of the speed controller as I_{ar} . The block diagram of the field current controller is shown in Fig.8 where:

$$D_f(z) = K_{pf} + K_{if}T_f \left[\frac{z}{z-1} \right] = K_{df} \left[\frac{z - z_{df}}{z-1} \right] \quad (33)$$

Although the field power circuit is designed with a current rating equal to that of the series field current winding, instead of the shunt field winding, the total losses of field circuit is greatly reduced and it is found practically that the speed of response in the two cases is comparable.

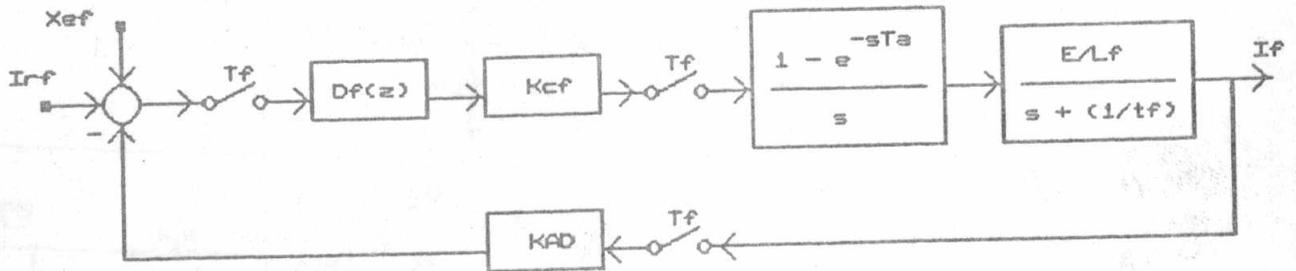


Fig.8 The Field Controller

Substituting the numerical values of field circuit $R_f = 6.7 \Omega$, $L_f = 0.268 \text{ H}$, and $\tau_f = 40 \text{ ms}$, the closed loop transfer function can be expressed as:

$$C_f(z) = A_f \left[\frac{z - z_{df}}{A_{2f}z^2 + A_{1f}z + A_{0f}} \right] \quad (34)$$

where: $A_f = K_{cf} K_{df} \left[\frac{E}{R_f} \right] \cdot [1 - e^{-T_f/t_f}]$; $A_{2f} = 1$

$$A_{1f} = K_{ADf} K_{cf} K_{DF} \left[\frac{E}{R_f} \right] \cdot [1 - e^{-T_f/t_f}] - (1 + e^{-T_f/t_f})$$

$$= 2.2243 \times 10^{-3} K_{df} - 1.8137$$

$$A_{0f} = -K_{cf} K_{df} \left[\frac{E}{R_f} \right] \cdot [1 - e^{-T_f/t_f}] + (1 + e^{-T_f/t_f})$$

$$= -1.1121 \times 10^{-3} K_{df} z_{df} + 0.81372$$

The stability boundary is :

$$0 < K_{df} < 2530 \quad ; \quad K_{df} - K_{df} z_{df} > 0$$

$$K_{df} + K_{df} z_{df} < 3268 \quad \text{These are reduced into:}$$

$$0 < K_{pf} < 1634 \quad \text{and} \quad K_{if} T_f > 0$$

The Measurement of the Currents

The armature current is measured, as shown in Fig.1, by sensing the voltage drop across a .02 Ω shunt in series with negative terminal of armature power circuit. The voltage appearing across shunt can be expressed as:

$$V_{shunt} = I_a \times R_{shunt} \times D_a \tag{34}$$

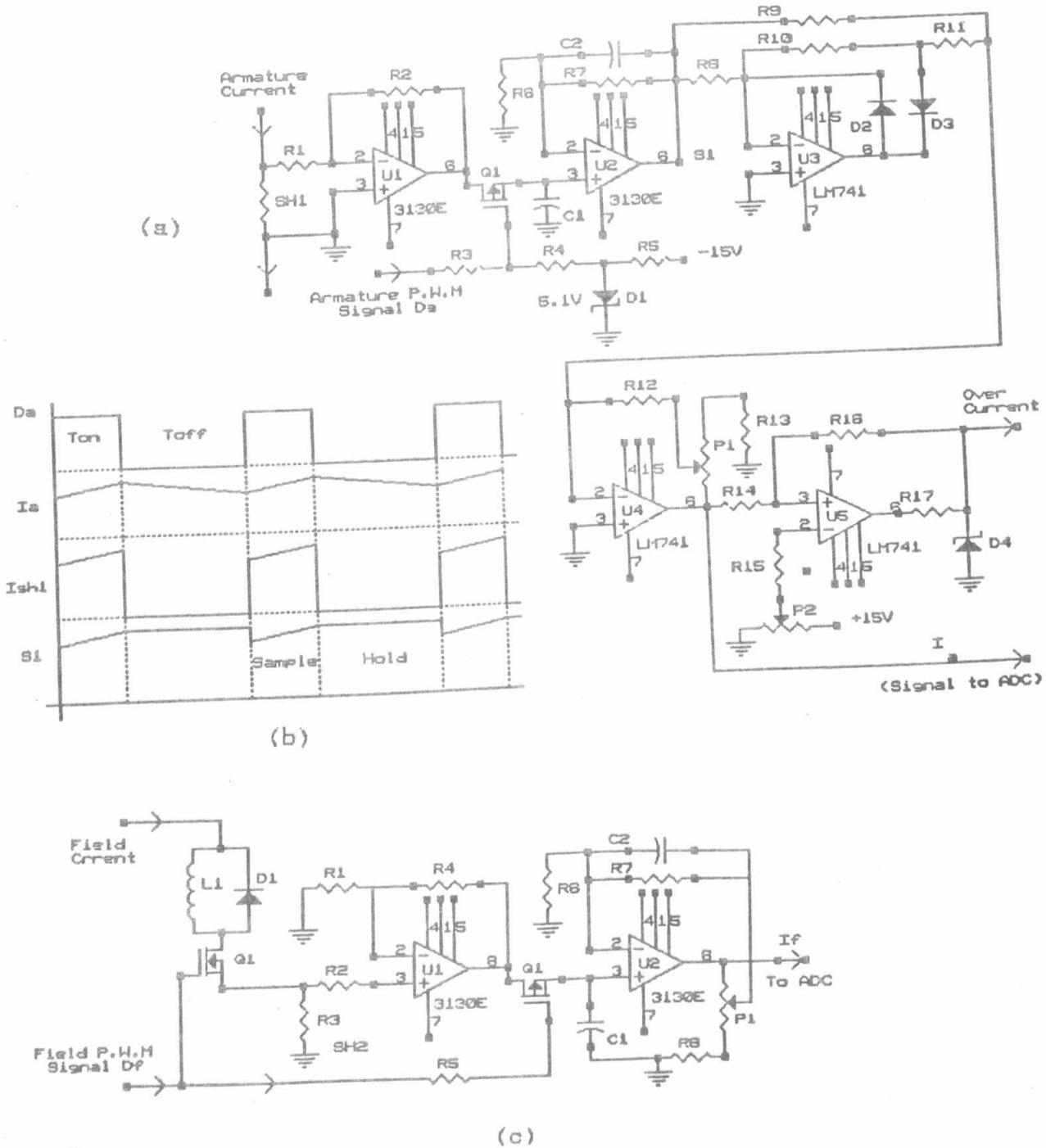


Fig.9 The Current Measurement

The analog switch Q1 in Fig.9.a, together with the operational amplifier U2 act as a Sample and Hold circuit, during the ON period Q1 is closed and the circuit output is proportional to I_a , during Off period Q1 (Fig.9.b) is opened holding the output at constant value till the next cycle. Since the chopper frequency is considerably high (31.125kHz), the armature leakage reactance greatly reduces the ripples in the current. The armature current measuring circuit gives the absolute value of current whatever its sign is. The comparator U5 is used to check if the armature current exceeds limit. Output of this comparator is connected to control system such that it generates a NMI to tell the MPU to take protective steps against the over current condition. The field current measuring circuit uses the same principle explained above, and it is shown in Fig.9.c.

IV. EXPERIMENTAL RESULTS AND CONCLUSION

These will be given in the second part (Part II) of the study where the complete structures of the used Software and the Hardware are implemented and the whole system is tested.

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